# A New Solar Position Sensor Using Low Cost Photosensors Matrix for Tracking Systems

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*Abstract:* - A solar tracking system is a machine that is designed for the orientation of different solar energy receivers, photovoltaic arrays or thermal receivers. The paper proposes a new solution of the solar position sensor used in tracking system control. The new matrix solar sensor (MSS) is an economic device but in the same time MSS advantages are the robustness and the accuracy. An actual tracking system that uses the new MSS achieves a position accuracy better then 1°. The main innovative concept of the new MSS is to use the ancient concept of solar clock to digitize the data set provided by the photosensors matrix area. The new MSS uses eight ordinary photo-resistors. These sensors working principle is to operate in two extreme conditions: in bright area or in shaded area. This binary principle simplifies the operating of the tracking control device. *Key-Words:* - solar energy, tracking systems, optimum receiver position, photo-resistors, matrix sensor, digital comparator.

# 1 Introduction

Renewable energy is a key concept in the 21<sup>st</sup> century. The catastrophic climate change, increasing energy needs in emerging economies (China, India), exhaustion of the classical energy resources and a secure energy supply are the principal features that must be taken into account when attempting to present the future of the world in an optimistic vision. The man-made changes of concentration of carbon-dioxide, methane and nitrous oxide are the principal factors that motivate interest in increasing renewable energy production. The main topics include photovoltaic, solar thermal energy, geothermal energy, wind energy, wave energy, bio-energy, hydrogen technologies and solar-assisted

cooling systems. Renewable energy involves efficient energy collection, efficient storage and transport and efficient energy conversion. To achieve an efficient solar energy capture imposes important requirements: concentrators, three tracking systems and MPP (Maximum Power Point) trackers. Simultaneous goals of the solar energy systems design are the robustness of the equipments and the cost of produced energy. The solar energy can be divided into photovoltaic technology and thermal solar technology. If a device combines photovoltaic and thermal collectors it is possible to increase the efficiency of the collectors.[1] Both requirements of high quality electrical insulation and high quality thermal conduction needed to be considered. The absorber collector design is shown in Figure 1.



Fig.1. The absorber collector design

The major advantages and disadvantages of PV systems are listed in Table 1.

Advantages of photovoltaics	Disadvantages of photovoltaics			
Fuel source is vast and essentially infinite	Fuel course is diffuse			
No emissions, no combustion or radioactive fuel for disposal (does not contribute perceptibly to global climate change or pollution)	(sunlight is a relatively low density energy)			
Low operating costs (no fuel) No moving parts Ambient temperature operation	High installation costs			
High reliability in modules (>20 years)	Poorer reliability of auxiliary (BOS)			
Modular (small or large increments)	elements including storage			
Quick installation				
Can be integrated into new or existing building structures	Lack of widespread commercially available			
Can be installed at nearly any point of use	system integration and installation so far			
Daily output peak may match local demand				
High public acceptance	Lack of economical efficient energy storage			
Excellent safety record				

Table 1. The advantages/disadvantages of PV systems

Some of disadvantages listed in Table 1, like relatively low density of the solar energy are corrected by the new PV technologies. The primary reason for using concentrators is to be able to use less solar cell material in a PV system. Concentrator systems increase the power output while reducing the size and numbers of cells needed. An additional advantage is that the cell's efficiency increases under concentrated light. Several challenges exist to using concentrators. The required concentrating optics are significantly more expensive than the simple covers needed for flat-plate solar systems, and most concentrators must track the sun throughout the day and year to be effective. Thus, achieving higher concentration ratios means using not only expensive tracking mechanisms, but also,

more precise controls than those of flat-plate systems with stationary structures. High concentration ratios also introduce a heat problem. When excess radiation is concentrated, so is the amount of heat produced. Cell efficiencies decrease as temperatures increase, and higher temperatures also threaten the long-term stability of solar cells. Therefore, the solar cells must be kept cool in a concentrator system. A concentrator photovoltaic panel (CPV) is depicted in Figure 2.



Fig.2. The concentrator photovoltaic panel

Absorber collectors systems (see Fig.1) are used to keep cool the concentrator photovoltaic panels. The economical analysis of the solar energy systems is an important element of the project. It is important to use the life cycle cost analysis (LCCA)[2].

There are two reasons to use an LCCA:

-to determine the most cost-effective design system -to compare different options

There are different initial investments costs and different operating, maintenance and repair costs. It is possible to obtain an economical performance of the solar energy system is the design can cat not only the initial investment cost but also the operating, maintenance and repair costs.

For all solar energy systems design, the first task is to determine the load. Then the amount of system storage needs to be determined. Some solar energy systems (photovoltaic or thermal) will not need storage systems or will have minimal storage systems. In the case of small size or medium size PV systems, the battery systems are the economically storage solution. In the case of medium and large power PV systems the storage systems use flywheel storage systems or Sodium-Sulfur battery.

The performance of the commercial Na-S battery bank is:

-capacity: 20-250 kW per bank -efficiency: 87% -lifetime: 2500 cycles (at 100% depth of discharge DOD) or 4500 cycles (at 80% DOD)

After battery selection must be determined the size of the PV panels and the electronic devices of the system: DC/DC, inverter, charge controller, maximum power point (MPP), tracking system are selected. The components include the mounting materials for the panels, the wiring, switches, fuses, protections and monitoring instrumentation. The social costs (greenhouse gas and pollution impact) are almost zero. When we replace the conventional power plants with renewable energy power plants the cost of the conventional fuel that saved is a part of the social cost. The quantity and the cost of the saved fuel are:

$$M_{fuel} = \frac{1}{LHV} \left( \sum_{i=1}^{t} \frac{X \cdot Y_i}{\eta_{ii}} \right) \quad [ton] \tag{1}$$

$$C_{fuel} = \frac{C_p}{LHV} \left( \sum_{i=1}^{t} \frac{X \cdot Y_i}{\eta_{ii}} \right) \quad [CU]$$
(2)

where:

 $M_{fuel}$  = quantity of fuel

 $C_{fuel} = \text{cost of fuel}$ 

*LHV* =lower heating value of fuel used at the input of conventional unit in [kJ/kg]

*X*=power in MW with is replaced by photovoltaic units

 $Y_i$ =percentage of full rated capacity with is generated by PV unit for a particular hour

 $C_p = \text{cost of fuel in currency unit/ton}$ 

 $\eta_{ii}$  = efficiency of conventional generation unit

The real and expected growth of solar (photovoltaic and thermal) energy has already applied some pressure on the supply of the most specialized types of motion components. Suppliers of drive systems and mechanical components have reported a huge upswing in interest from OEMs who integrate solar power systems with Sun-tracking capabilities, because the positioning accuracy of a solar tracking system improves the energy collection of the whole system. There are different model of trackers.[3] They are classified according to the movements they perform:

-single axis solar trackers

-horizontal axis solar trackers

-vertical axis (azimuth) solar trackers

-dual axis solar trackers

The dual axis solar trackers are classified under the following headings:

-dual axis polar mount trackers -dual axis parallel cinematic trackers -rotating platform dual axis trackers

rotating platform dual axis trackers

A commercial dual axis polar mounted solar tracker is shown in Figure 3.



Fig. 3. Dual axis polar mounted solar tracker

Dual axis rotating platform solar trackers are used to coordinate the movement of a great number of solar collectors (PV or thermal). It is a robust and simple tracker system. In Figure 4 is depicted a rotating platform tracker.



Fig.4. The rotating platform solar tracker

The double movement of the Earth, around the Sun and the rotation about its inclined polar axis is shown in Fig.1.



Fig.1. The Earth's orbit around the Sun, showing the inclined axis of diurnal rotation

The angle of deviation of the Sun from directly above the equator is the declination angle,  $\delta$ . The mathematical expression of the declination in accord with the day (n) of the year is [4]:

$$\delta = 23.45^{\circ} \cdot \sin\left[\frac{360 \cdot (n-180)}{365}\right]$$
(1)

In (1), those angles north of the equator are positive and those south of the equator are negative.

The zenith angle,  $\vartheta_z$ , is the angle between the Sun and a line perpendicular to the Earth's surface (the zenith line). Since the Sun is directly overhead on the first day of summer, at noon, on the tropic of cancer, the zenith angle is given by:

$$\vartheta_{z} = \phi - \delta \tag{2}$$

where  $\phi$  =the latitude in degrees

The relations between the deviation angle and the zenith angle are illustrated in Fig.2.



Fig.2. The relation between deviation and zenith angles

If we assume the distance from the Earth to the Sun to be constant, the position of the Sun can be completely specified by two coordinates: the solar altitude and the azimuth. The solar altitude  $\alpha$ , is the complement of the zenith angle  $\vartheta_z$ . The solar altitude represents the angle between the incident solar beam and the horizon, in the plane determined by the Sun and the zenith.

The necessary pair of coordinates is illustrated in Fig.3. The angle  $\psi$  is the azimuth angle and measures the angular deviation of the Sun from due south. The azimuth angle is zero at solar noon and increases toward the east. In some papers, the azimuth angle is referenced to north and in this case the solar noon appears at  $\psi = 180^{\circ}$ . The angle  $\omega$  describes the angular displacement of the Sun from solar noon in the plane of apparent travel of the Sun. This angle is useful but redundant in the system of solar coordinates.



Fig.3. Illustrating the pair of coordinates necessary in a complete determination of the solar position

The relationships between  $\psi$  and  $\alpha$  may be determined, but are difficult to visualize. For this reason it is convenient to plot the values of angles  $\psi$  and  $\alpha$  for specific latitudes and days of the year. Fig.4 shows a set of plots for the case of a latitude  $30^{0}$  N [5]



Fig.4. The values of angles  $\psi$  and  $\alpha$  for a latitude of  $30^{\circ}N$ 

Irradiance is the measure of the power density of Sunlight. Is an instantaneous quantity and is measured in  $W/m^2$ . The irradiance received by the Earth from the Sun at the top of the atmosphere is constant, equal to 1367  $W/m^2$ . The irradiance is reduced after passing through the atmosphere to an approximate value of 1000  $W/m^2$ . The direct normal solar irradiance is given by:

$$I_{t,h} = I_{b,n} \cdot \cos \vartheta_z + I_{d,h} \tag{3}$$

 $I_{b,n}$  = the irradiance coming directly from the Sun

 $I_{d,n}$  = the diffuse radiation

The clear-day model of direct normal solar irradiance using the U.S. Standard Atmosphere (Hottel model) is given by [5][7]:

$$I_{b,n} = I_0 \left( a_0 + a_1 e^{-k \frac{1}{\cos \theta_z}} \right)$$
(4)

where:

 $I_0$  = the extraterrestrial radiation

 $a_0, a_1, k$  =parameters

$$a_a = 0.4237 - 0.00821(6 - A)^2$$
  

$$a_1 = 0.5055 - 0.00595(6.5 - A)^2$$
  

$$k = 0.2711 - 0.01858(2.5 - A)^2$$

To completely specify the position of the sun it is three coordinates are needed: the distance between the sun to the Earth, the solar altitude  $\alpha$  and the azimuth angle  $\psi$  [8]. If one assumes the suncollector distance from the sun to be constant, the sun position can be specified using two coordinates. The complement of the zenith angle  $\theta_z$  is the solar altitude  $\alpha$  and represents the angle between the horizon and the incident solar beam in a plane determined by the zenith and the sun.

The azimuth angle  $\psi$ , is the angular deviation of the sun from directly south (or north in some papers). The azimuth angle is zero at solar noon and increase toward the east. If the azimuth is referenced to north, the solar noon appears at  $\psi = 180^{\circ}$ .

The hour angle  $\omega$  is the difference between noon and the desired time. We assume  $360^{\circ}$  rotation in 24 hours:

$$\omega = \frac{12 - T}{24} \cdot 360^\circ = 15(12 - T)^\circ$$

The all angles used in the determination of the Sun position are shown in the Figure 5.



Fig.5. Complete angles system used for the determination of the Sun position

More solar energy is collected by the end of the day if the solar receivers (photovoltaic or thermal) are installed with a tracker system. For a planar receiver, the receiver plane must be perpendicular to the direction of the solar flux irradiance.

The solar energy collected is proportional to the cosine of the angle between the incident beam and the normal of the plane of the collector, as shown in Figure.6. [7]



Fig.6. The collected solar energy

There are two types of Sun trackers:

- A one-axis tracker: follows the Sun from east to west during the day
- A two-axis tracker: maintains the receiver surface perpendicular to the Sun and allows collection of the maximum possible amount of energy.



Fig.7. The collected energy by fixed and tracker receivers

The collected Sun energy improves by 40%-50% using tracking systems (35%-42% by E-W tracker and 5%-8% by N-S tracker). The improvement in the collected energy varies during the year. The energy collected for a fixed receiver, an adjusted one (one axis) and a two-axis tracker is depicted in Fig.7. In the summer time and in a dry climate, the additional energy is approximately 50% but in the winter time, the additional energy is only 20%.

#### **2** Sun tracker control systems

The ideal Sun tracker system is one that allows tracking along two axes, by means of a simple mechanism, with is manufactured with commonly used industrial components (mechanical or electrical) and with requires low-level of maintenance. Sun tracker control systems use two major strategies:

 the sensor control: the Sun's position is specified using two coordinates and the optimum-optimum position is estimated using photosensors signals. • the sensorless control: the optimum-optimum values of the Sun coordinates are calculated for the any moment of the day, during all year, in the case of a fixed location of the solar collector. Using this data base, the tracking system moves the collector to the optimum-optimum position. The data base must be calculated for every geographical location of the Sun collector.

Figure 8 shows the cumulative collected energy for fixed collector and tracking collector:



Fig.8. The cumulative collected energy

The Sun tracking controllers were developed following the classical control system closed-loop approach by integrating a Sun sensor able to provide pointing-error signals, one per tracking axis. This in turn generates actuator correction movements. Each Sun sensor comprises a pair of phototransistors, generate different photocurrents whenever the sensor is not aligned with the local Sun vector. The phototransistors may be mounted on tilted planes in order to increase sensitivity. In some applications, the shading devices are provided. Some types of Sun sensor are shown in Fig.7.[8]



Fig.7. Various types of Sun sensor

A classical sensor for one Sun position coordinate uses a pair of phototransistors mounted on titled planes. The angle of tilt of the plane is  $90^{\circ}$ . The device is encased to provide protection. The construction of the sensor is illustrated in Fig.8.



Fig.8 A classical Sun sensor

There are two important problems: for each sensor the phototransistors must be quasi-identically, because the measurement method assumes identically currents in the case of identically irradiance. If the characteristics of the phototransistors are not identically, the position of the collector is not the optimal position. The second problem is the ageing semiconductor effect. The ageing phototransistors effects and the accidentally greasing of the sensor box induce a faulty function of the tracking system.

#### 2.1 Proposed new Sun sensor design

The problems of the classical Sun sensor may be solved by the proposed new type of matrix Sun sensor (MSS). In the previous solutions, each tracking direction is controlled by using a Sun sensor made by a pair of phototransistors. This proposal is for a single matrix Sun sensor MSS which controls both axes of the tracking system. The inspiration for the MSS is the antique solar clock. MSS comprises 8 photo-resistors and a cylinder. The location of the cylinder is at the centre of the matrix structure of the photo-resistors, which are evenly distributed around a circle. If the position of the collector is not optimum, the shadow of the cylinder covers one or two photo-resistors. The photo-resistors are mounted circular, around the cylinder, as shown in Fig.9.

The photo-resistor is a high resistance semiconductor whose resistance decreases with increasing intensity of incident light. It is a light dependent resistor device. The basic principles of the photoconductive effect are:

-directly beneath the conduction band of the CdS crystal is a donor level and there is an acceptor level above the valence band. In darkness, the electrons and holes in each level are almost crammed in the crystal and the CdS is a high resistance

-when light illuminates the CdS cell, the electrons in the valence band are excited into the conduction band. This creates pairs of free electrons in conduction band and free holes in the valence band, increasing the conductance



Fig.9 The photoresistors matrix

The diameters of the CdS cell are: 3mm, 4mm, 5mm, 7mm, 12mm, 20mm. For the particular case of the matrix application there are uses cells with 3mm diameter. The structure and the lead terminals are depicted in Figure 10.



Fig.10. The lead terminals of the CdS cell

The illuminated resistance, the dark resistance and the response speed vary with the conditions to with the CdS cell has previously been exposed. This effect is called the light history effect. In general, when the CdS cell is kept in darkness for a long time, its illuminated resistance will be lower compared to a cell kept at a bright level. In the particular case of the matrix application the light history effect doesn't influence the accuracy of the matrix solar sensor. In order to reduce the light history effect, the CdS cells can be used after being exposed to light for several minutes. The principle of the Sun tracker assures this condition. The change of the CdS cells resistance with ambient temperature depends on the light level. The temperature characteristic is illustrated in Figure 11.



Fig.11. Temperature characteristic of CdS cell

The cadmium sulphide cell, CdS cell, is the most inexpensive type of photo-resistor (the price of a CdS cell is 10-12 eurocents). This is why the proposed implementation of MSS uses CdS cells. There are two important observations about MSS operation:

- the classical control system closed-loop approach is substituted by a simple digital circuit
- the MSS works in a digital kind of way: each CdS cell has two states, illuminated state and shaded state

#### 2.1.1 MSS sizing and additionally circuits

The topology and working principle of the proposed MSS respect the antique solar clock. Each CdS cell corresponds to a direction: N, NE, E, SE, S, SW, W and NW (see Fig.9). The MSS logic operation is:

- if the N CdS cell is shaded, the N-S actuator moves in the opposite direction, southward.
- if the N-E CdS cell is shaded, both actuators must move in the opposite direction: N-S actuator moves southward and E-V actuator moves westward.
- if the E CdS cell is shaded, the E-V actuator moves in the opposite direction, westward.
- if the S-E CdS cell is shaded, both actuators must move in the opposite direction: N-S actuator moves northward and E-V actuator moves westward.
- if the S CdS cell is shaded, the N-S actuator moves in the opposite direction, northward.

- -if S-W CdS cell is shaded, both actuators must move in the opposite direction: N-S actuator moves northward and E-V actuator moves eastward
- if the W CdS cell is shaded, the E-V actuator moves in the opposite direction, eastward.
- if the N-W CdS cell is shaded, both actuators must move in the opposite direction: N-S actuator moves southward and E-V actuator moves eastward.

The difference between a shaded CdS cell and a lighted CsS cell is recognised using the curren values of the cell resistance. The electronic circuit is a simple, robust, low-cost voltage divider circuit. Two circuit variants are illustrated in Fig.10:

- the output voltage increases when the irradiance increases (Fig.10 a)
- the output voltage decreases when the irradiance increases (Fig.10 b)

For the a,b circuits, the output voltages are:

$$V_{out} = \frac{R_2}{R_2 + R_1} \cdot V_{cc} \tag{5}$$

$$V_{out} = \frac{R_1}{R_2 + R_1} \cdot V_{cc} \tag{6}$$



Fig.10 Voltage divisor circuits for CdS cell

The MSS uses the divider circuit type a. The resistance  $R_2$  is:

$$R_2 = \sqrt{R_{1dark} \cdot R_{1bright}} \tag{7}$$

It is necessary to measure the resistance of all CdS cells with respect to the level of illumination. The values are in the Table 2. The percentages represent the levels of lighted area of a cell.

	Ν	NW	W	SW	S	SE	Е	NE
100%	46.4	40.6	42.8	41.4	39.4	40.4	41.8	42
75%	62.8	55	61.6	54.6	50.2	53	57.4	56.6

50%	78.4	78.2	77	65.6	67.4	70.2	73.4	80.6
25%	110.4	101.8	116.6	99.4	110.6	112.2	112.6	110
0%	263	171.2	205.2	186.6	234.2	254	293	307.6

Table 2. The values of CdS cell resistance Using the relationship for  $R_2$ , the computed values are in Table 3:

	Ν	NW	W	SW	s	SE	Е	NE
<b>R</b> <sub>2</sub>	110.5	83.4	93.7	87.9	96	101.3	110.7	113.7
	Tab	le 3. '	The v	alues	of	compu	ited R	,

For the same condition of illumination the values of the CdS cell resistances are different, according to their position relative to the central cylinder. The variations are important. The classical Sun position sensor cannot work under these conditions. The variations referred to the mean values are shown in Table 4.

	$\mathcal{E}\%$							
	Ν	NE	Е	SE	S	SW	W	NW
100%	10.9	-2.9	2.3	1.1	-5.8	-3.4	-0.1	0.4
75%	11.3	-2.5	9.2	-3.2	-11	-6	1.8	0.3
50%	6.1	5.8	4.2	-11.2	-8.8	-5	-0.7	9
25%	1.1	-6.8	6.8	-8.9	1.3	2.7	3.4	0.7
0%	9.9	-28.5	-14.2	-22	-2.1	6.1	22.4	28.5

Table 4. The errors of CdS cells resistances

Using data from Table 3, the value of  $R_2$  is 120  $\Omega$ . The output voltages, in accord with relationship (5), using the value of  $R_2=120 \Omega$ , the reference voltage  $V_{CC}=5V$  and the measured resistances of the CdS cells from Table 2, are presented in Table 5 (all values are given in V).

	U <sub>N</sub>	U <sub>NE</sub>	U <sub>E</sub>	U <sub>SE</sub>	Us	U <sub>SW</sub>	Uw	U <sub>NW</sub>
100%	3.60	3.73	3.68	3.71	3.78	3.74	3.70	3.71
75%	3.28	3.42	3.30	3.43	3.52	3.46	3.38	3.39
50%	3.02	3.02	3.04	3.23	3.20	3.15	3.10	2.99
25%	2.60	2.70	2.53	2.73	2.60	2.58	2.57	2.60
0%	1.56	2.06	1.84	1.95	1.69	1.60	1.45	1.40

Table 5. The output voltages of CdS divisor circuits

The voltage variations of all CdS circuits, for the same illumination conditions are given in Table 6. The variations are reduced (the effect of the presence of  $R_2$ ) but there are still significant.

	€%	$\mathcal{E}\%$	$\mathcal{E}\%$	$\mathcal{E}\%$	€%	$\mathcal{E}\%$	$\mathcal{E}\%$	$\mathcal{E}\%$
	Ν	NE	Е	SE	S	SW	W	NW
100%	-2.9	0.5	-0.1	0	1.9	0.8	-0.2	0
75%	-3.5	0.6	-2.9	0.9	3.5	1.8	-0.6	-0.3
50%	-2.5	-2.5	-1.9	4.2	3.2	1.6	0	-3.5
25%	0	3.8	-2.7	5	0	-0.8	-1.1	0
0%	-8.2	21	8.2	14.7	-0.6	-5.9	-14	-17

Table 6. Voltage variations of the CdS divisor circuits

It is possible to obtain Sun sensor accuracy smaller then 1° if MSS uses a "binary" technique: MSS compares illuminated CdS cells with shaded CdS cells. This operation is performed using eight comparator circuits. Each comparator uses 2 signals: the output voltage of the divider circuits and the reference voltage. The reference voltage is selected in the illumination range 0%-30%, from Table 4. If the selected reference voltage is 2V-2.5V, MSS and the comparators block generate a binary output: 1 if the CdS cell is lighted and 0 if the CdS cell is shaded. The output signals of the comparator block are: N, NE, E, SE, S, SW, W, NW. The digital behavior of the sensor matrix and the comparators block is given in Figure 11.



Fig.11. The digital behavior of the matrix sensor

The MSS discrimination is defined by the dimensions of the central cylinder. If the accuracy is fixed 1°, the length of the cylinder may be determined by a simple trigonometric relationship, as shown in Fig.12. When the deviation of the optimum position of the collector is 1<sup>°</sup>, the shadow of the cylinder cast must be 3mm (the dimension of the CdS cell). The length of the cylinder is thus:

$$h = \frac{s}{tg\alpha} = \frac{3}{tg1^0} \approx 17 \ [cm] \tag{7}$$



#### 2.1.2 MSS digital control

The binary character of the new MSS suggests a simple digital control of the tracker actuators. Circuit design starts with the truth table, see Table 7, where the input signals are N, NE, E, SE, S, SW, W, and NW (these signals are the output digital set of the comparator block) and the output signals,  $D_N$ ,  $D_E$ ,  $D_S$ ,  $D_W$ , determine the start/stop functions of the N-S and E-W actuators.

N	NE	E	SE	S	SW	W	NW	D <sub>N</sub>	D <sub>E</sub>	Ds	$\mathbf{D}_{\mathbf{W}}$
1	0	0	0	0	0	0	0	0	0	1	0
1	1	0	0	0	0	0	0	0	0	1	1
0	1	0	0	0	0	0	0	0	0	1	1
0	1	1	0	0	0	0	0	0	0	1	1
0	0	1	0	0	0	0	0	0	0	0	1
0	0	1	1	0	0	0	0	1	0	0	1
0	0	0	1	0	0	0	0	1	0	0	1
0	0	0	1	1	0	0	0	1	0	0	1
0	0	0	0	1	0	0	0	1	0	0	0
0	0	0	0	1	1	0	0	1	1	0	0
0	0	0	0	0	1	0	0	1	1	0	0
0	0	0	0	0	1	1	0	1	1	0	0
0	0	0	0	0	0	1	0	0	1	0	0
0	0	0	0	0	0	1	1	0	1	1	0
0	0	0	0	0	0	0	1	0	1	1	0
1	0	0	0	0	0	0	1	0	1	1	0
7	[abl	- 7	The	trut	h tal	hle o	of the	- digi	tal M	55 d	evice

table of the digital MSS device

For other combinations all outputs become zero. The shaded values are redundant. The optimized truth table is given in Table 8:

Ν	NE	E	SE	S	SW	W	NW	$\mathbf{D}_{\mathbf{N}}$	D <sub>E</sub>	Ds	$\mathbf{D}_{\mathbf{W}}$
1	0	0	0	0	0	0	0	0	0	1	0
0	1	0	0	0	0	0	0	0	0	1	1
0	0	1	0	0	0	0	0	0	0	0	1
0	0	0	1	0	0	0	0	1	0	0	1
0	0	0	0	1	0	0	0	1	0	0	0
0	0	0	0	0	1	0	0	1	1	0	0
0	0	0	0	0	0	1	0	0	1	0	0
0	0	0	0	0	0	0	1	0	1	1	0

Table 8. The optimized truth table

The output signals are:

 $D_N = SE + S + SW$ ,  $D_E = SW + W + NW$ ,

 $D_S=NE + N + NW$ ,  $D_W=NE + E + SE$ , (8) Practically, there are some forbidden input signals combinations. The shadow area can cover one, maximum two adjacent areas. The correct combinations are: N-NE, NE-E, E-SE, SE-S, S-SW. SW-W, W-NW, NW-W. All these combinations are in the digital relationships (8). The forbidden combinations are: N-E, N-SE, N-S, N-SW, N-W, NE-SE, NE-S, NE-SW, NE-W, NE-NW, E-S, E-SW, E-W, E-NW, SE-SW, SE-W, SE-NW, S-W, S-SW, SW-NW.

The digital circuit is a code convert type and is illustrated in Fig.13. The new MSS was tested in a solar parabolic dish thermal collector, at Politehnica University Timisoara, Romania (Fig.14).



Fig.13. Code converter circuit



Fig.14. A solar parabolic dish thermal collector using the new MSS

### **3** Conclusion

A new Sun sensor matrix structure is proposed. MSS uses low cost unsorted photoelements (CdS cells). CdS cell works in a digital way because MSS observe 2 extreme situations: lighted and shaded CdS cell resistances. The MSS discrimination is better than  $1^{\circ}$  and it is constructed using a cylinder mounted in the centre of the CdS matrix structure. The sensor is robust, low cost and determines a simple digital control of the actuators of the Sun tracker.

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